

The German Aerospace Center – DLR

Deutsches Zentrum für Luft- und Raumfahrt e.V





- DLR as large scale research facility:
 - is the largest **Science Center for Aerospace** in Germany with the main research areas: aviation, space, transport, energy and security
 - is the **German Space Agency**. In this role DLR manages the German space program on behalf of the government
 - Is one of the largest Project Administration for publicly founded projects in Germany
- DLR has about 10.000 employees in 54 institutions at 27 locations



DLR's research goals in the area of high altitude solar platforms



- DLR is developing a full scale technology demonstrator for high altitude solar platforms involving the program areas of aviation, space and security
- Goals of the development are:
 - Development of novel system concepts and technologies for the realization of robust and cost-efficient high altitude solar platforms
 - Development and testing of innovative sensor systems for earth observation. In particular high performance is combined with highly sophisticated lightweight construction
 - Development of operational strategies and mission scenarios to demonstrate the performance of high altitude solar platforms
 - Demonstration and flight testing of novel technologies, processes and sensor systems under real environmental conditions

DLR Project HAP-alpha

Project initiated in 2018 with a budget of around 30 M€

Currently 16 DLR Institutes all over Germany are involved:

Flight Systems FT,

Systemhaus Technik SHT,

Aeroelasticity AE,

Flight Guidance FL,

Remote Sensing Technology MF,

Atmospheric Physics PA,

Microwaves and Radar HR,

Networked Energy Systems VE,

Software Technology SC,

Aerodynamics and Flow Technology AS,

Communications and Navigation KN,

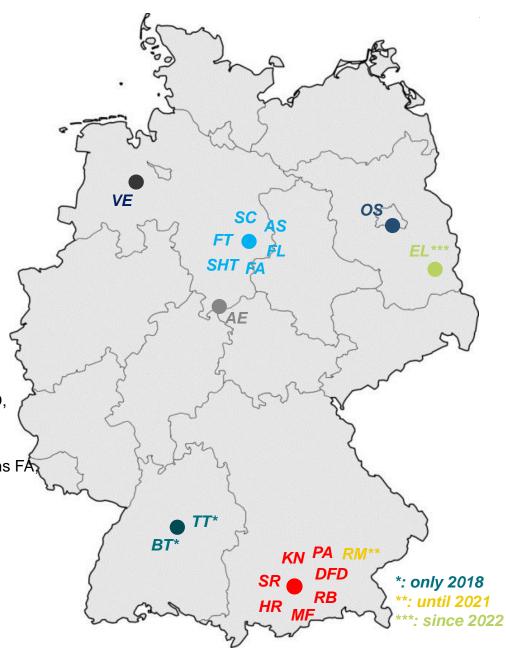
Optical Sensor Systems OS,

German Remote Sensing Data Center DFD,

Electrified Aero Engines EL,

Composite Structures and Adaptive Systems FA

System Dynamics and Control SR



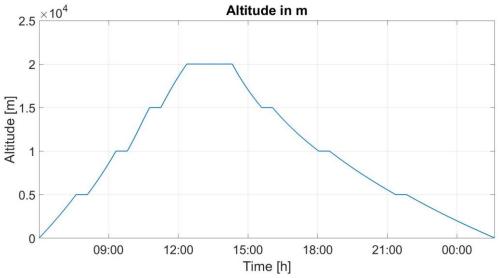
Technology Demonstrator HAP alpha

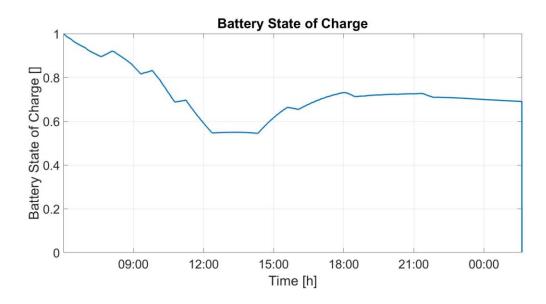


Research Mission Design



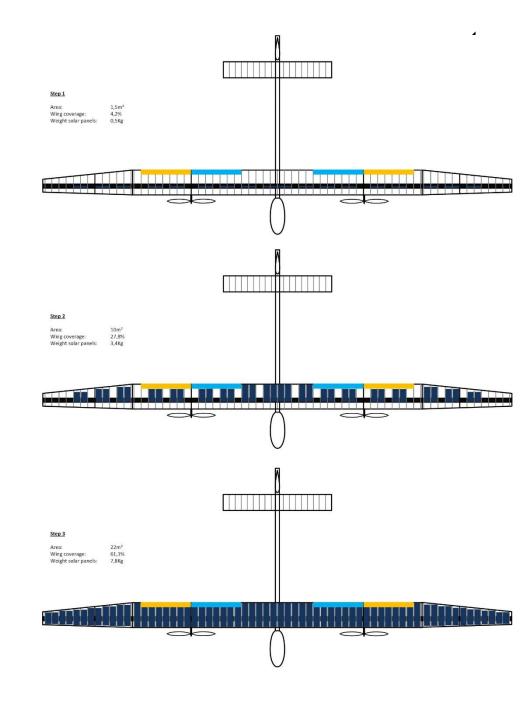
- We aim to operate the aircraft under real flight conditions
- The goal is to reach an altitude of 20 km and remain there for at least 2 h
- The aircraft is designed for several landings
- The aircraft is designed according to modified CS25 gust loads
- The aircraft shall serve as a testbed for critical system technology
- We want to operate payload systems at the target altitude
- We want to built up and advance our capabilities to conduct long term flight test
- We do not aim to develop a commercial product





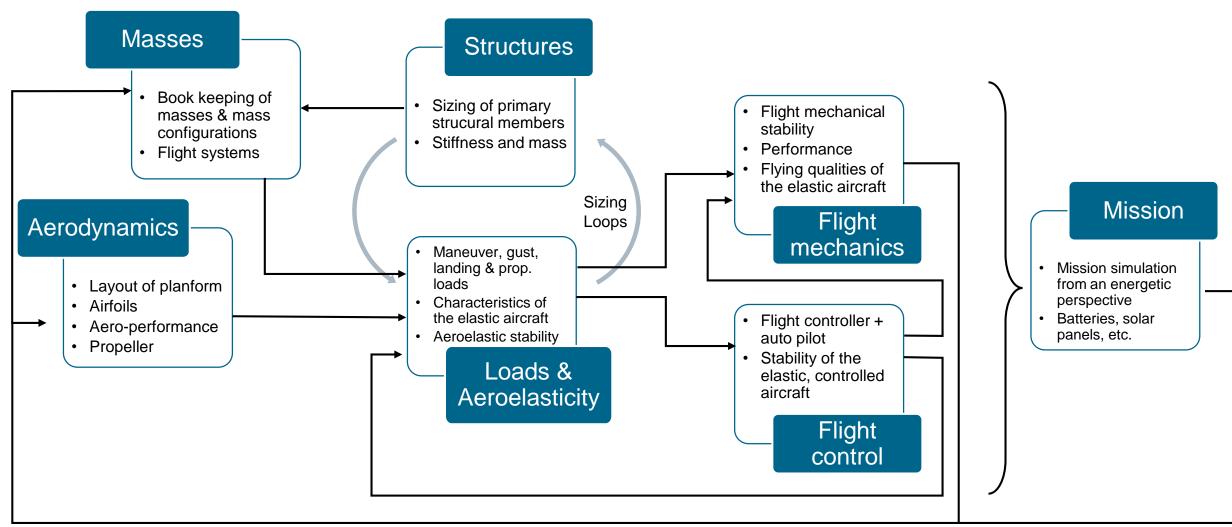
HAP alpha Variants

- As cost reduction we apply in the first steps only a small area of solar panels on the wing
 - Variant 1: ~ 1.5 m² Solar panels
 - Variant 2: ~ 12 m² Solar panels
 - Variant 3: ~ 22 m² Solar panels (at the moment no further consideration)
- Each variant differs in the area of applied solar panels only



Aircraft Design Loop





Drive Train



DLR

Solar Arrays

- Redundancy achieved via twice instantiation on aircraft level to satisfy system reliability and availability requirements
- Applied design principles of high cohesion and loose coupling to reduce system complexity
- Tailored to the special requirements in-house developments
 - Engine Control Unit
 - Power Distribution Unit
 - Battery System
 - Solar Power Tracker
- Distributed battery system throughout the wingspan utilizing standard cell geometry
- Engine is a modified COTS component



Propulsion System



Battery Unit



Power Distribution Unit

Avionics



Sensors & Actuation

Actuation Position
System Sensorics

Air Data System Logic

Flight Control
Computer

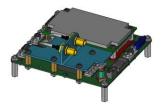
Communication

Command & Control Link

Payload Link

Transponder System

- High-Performance INS for payloads developed by DLR
- Redundant INS integrated into Flight Control Computer



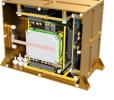
- In-house hardware and software
- Duplex system (COM/MON)
- Integrated INS and air data
- RTOS running Flight Control
 & Management
- HIL testing through ethernet connection

- High-rate downlink for observation data
- Flow control through C2 Link
- COTS hardware



- 6 control surfaces
- Special actuators with wide temperature range
- Low-EMI adaptation

- High precision, low weight
- Power redundancy
- Cooperation with industry partner



- COTS Components
- High-range S-Band link
- Up-/download



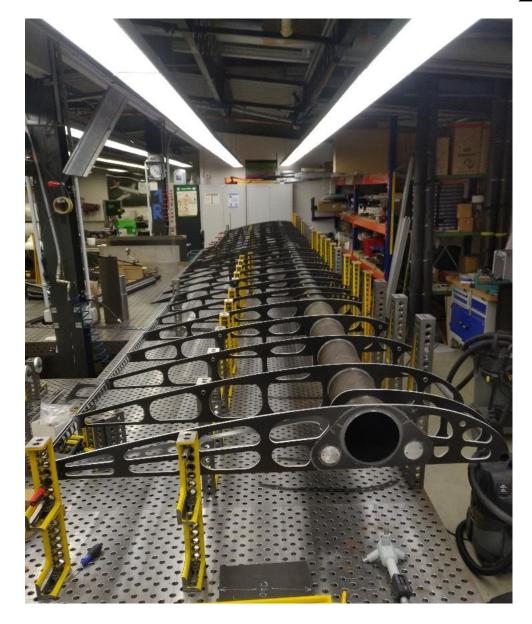
- Custom UPS
- ADS-B transponder



Impressions of the realization of the structure









HAP alpha prototype hardware testing







Prototype Tests include drive tests of motor and propeller as well as extensive temperature and pressure testing of all airborne hardware.

Where the Project is now



- By now the project is at the end of the detailed design phase:
 - Final aircraft configuration established
 - All system elements and items are designed
 - Prototypes of system hardware and software, in function and performance close to the final flight hardware, have been tested
 - Ground segment designed an under production
 - Critical Design Review planned for July in preparation

What to expect in 2023



- Transition to the overall system test phase:
 - Full scale wing Ultimate Load test
 - Manufacturing of the flight structure
 - Quality assurance activities for all flight hardware
 - Stepwise buildup and test to a complete HAP system
- Delivery and commissioning of the Ground Segment
- Extensive crew training
 - Remote pilot and safety pilot training with downscaled HAP alpha model aircraft
 - Overall crew training for flight operations



Andreas Bierig

Head of Department: Safety Critical Systems and Systems Engineering Institute of Flight Systems +49 531 295 2403

Andreas.Bierig@dlr.de

Florian Nikodem

Project Manager High Altitude Platform (HAP alpha) +49 531 295 3503

Florian.Nikodem@dlr.de

Dr. Steffen Niemann

Systems Engineer – Aircraft Structure +49 531 295 2966 Steffen.Niemann@dlr.de

